

# Optical Response Simulation for ASTRO-G Laser Reflector Array

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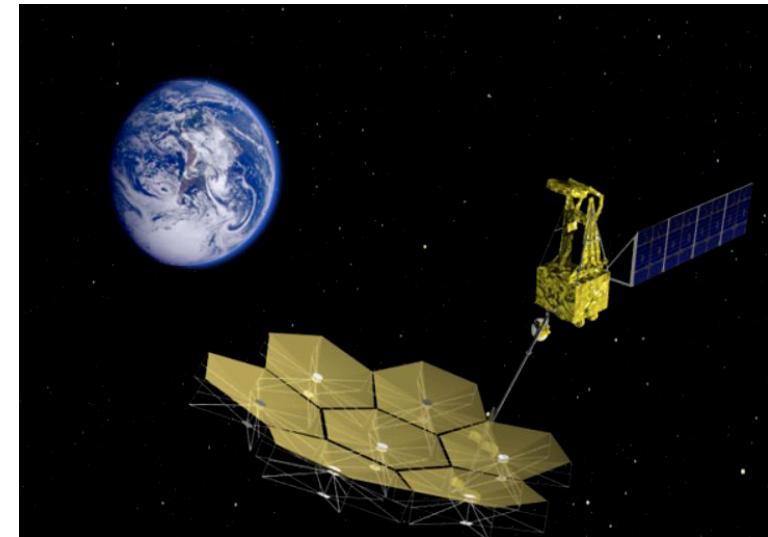
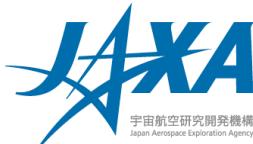
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National Institute of Information  
and Communications Technology



# ASTRO-G Highly Elliptic Orbit

## ASTRO-G (VSOP-2)

Launch: FY2012 (5-year mission)

9.6-metre antenna

Highly elliptic orbit: 1000 x 25000 km altitude

Observation bands: 8.4, 22 and 43 GHz

High frequency, High resolution and High sensitivity



## 4 x GPS (+Galileo?) Receiver

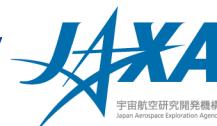
Effective <3000-5000 km only (1 hr per 7.5 hrs).

Sidelobe? One-frequency use?

## 1 x SLR Retroreflector Array ← This talk

Should be effective 1000 km to 25000 km

More than a cal/val instrument



# 4-D Simulation of CCR Response

## 4-Dimensional Function:

Angle of incidence and azimuth (2-D)

Velocity aberration (2-D)

## Software development for Single CCR Response (ongoing)

Language: C#

Input:

CCR Shape, Optical Index, Coat, Size, Recession, Dihedral angle

Laser wavelength, Polarisation

Output:

Far-field amplitude

Grid size: 2-deg for angle of incidence, 2- $\mu$ rad for velocity aberration

> 2 GB in ASCII Text, > 100 MB in Binary (NetCDF) file

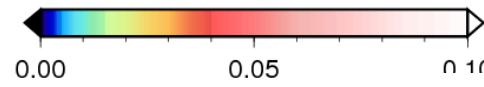
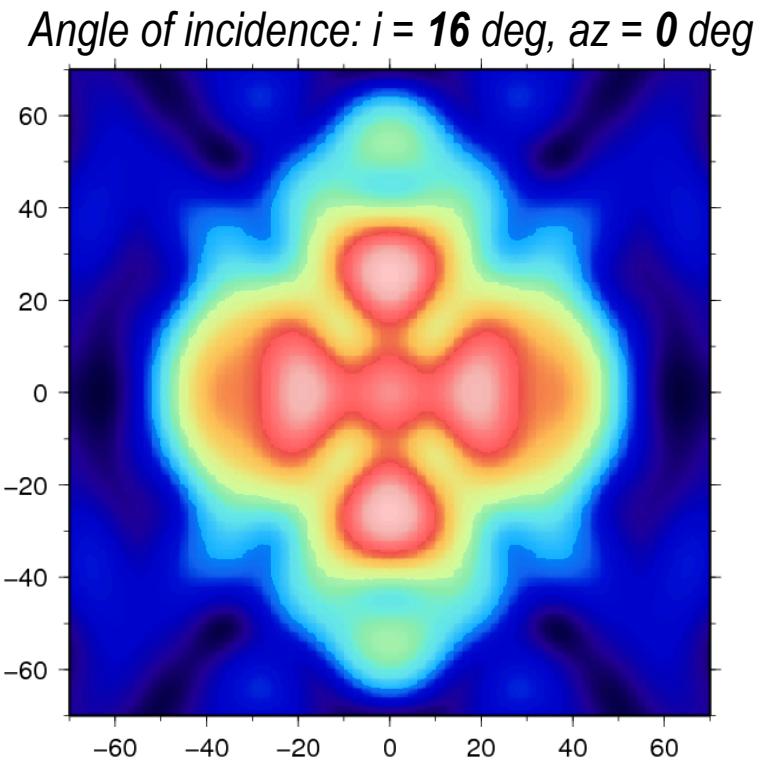
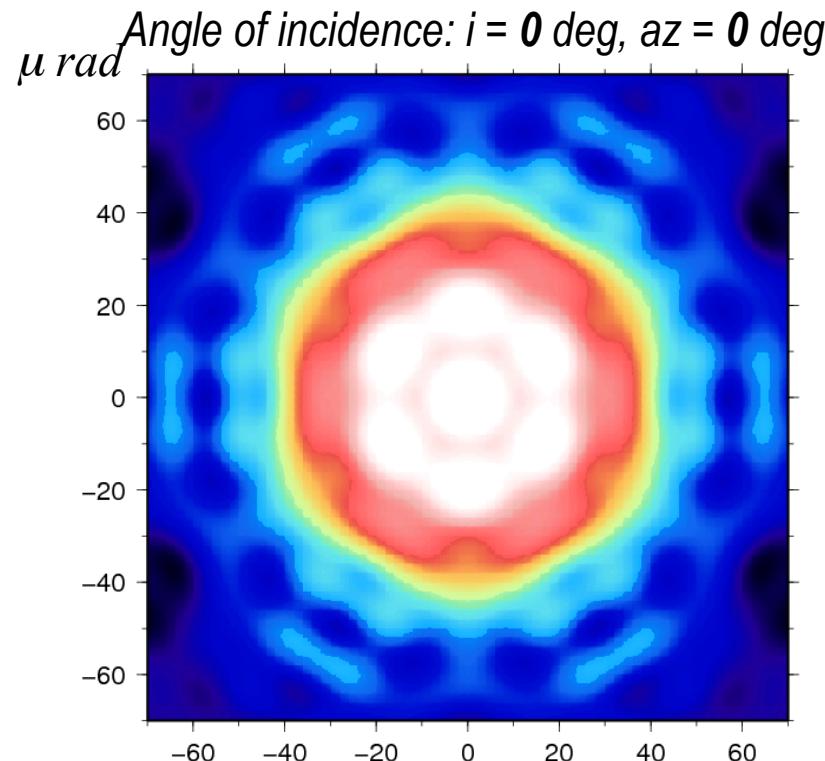
Computation time: 6 to 14 hours per reflector ... needs optimisation



# Far Field Diffraction Pattern (examples)

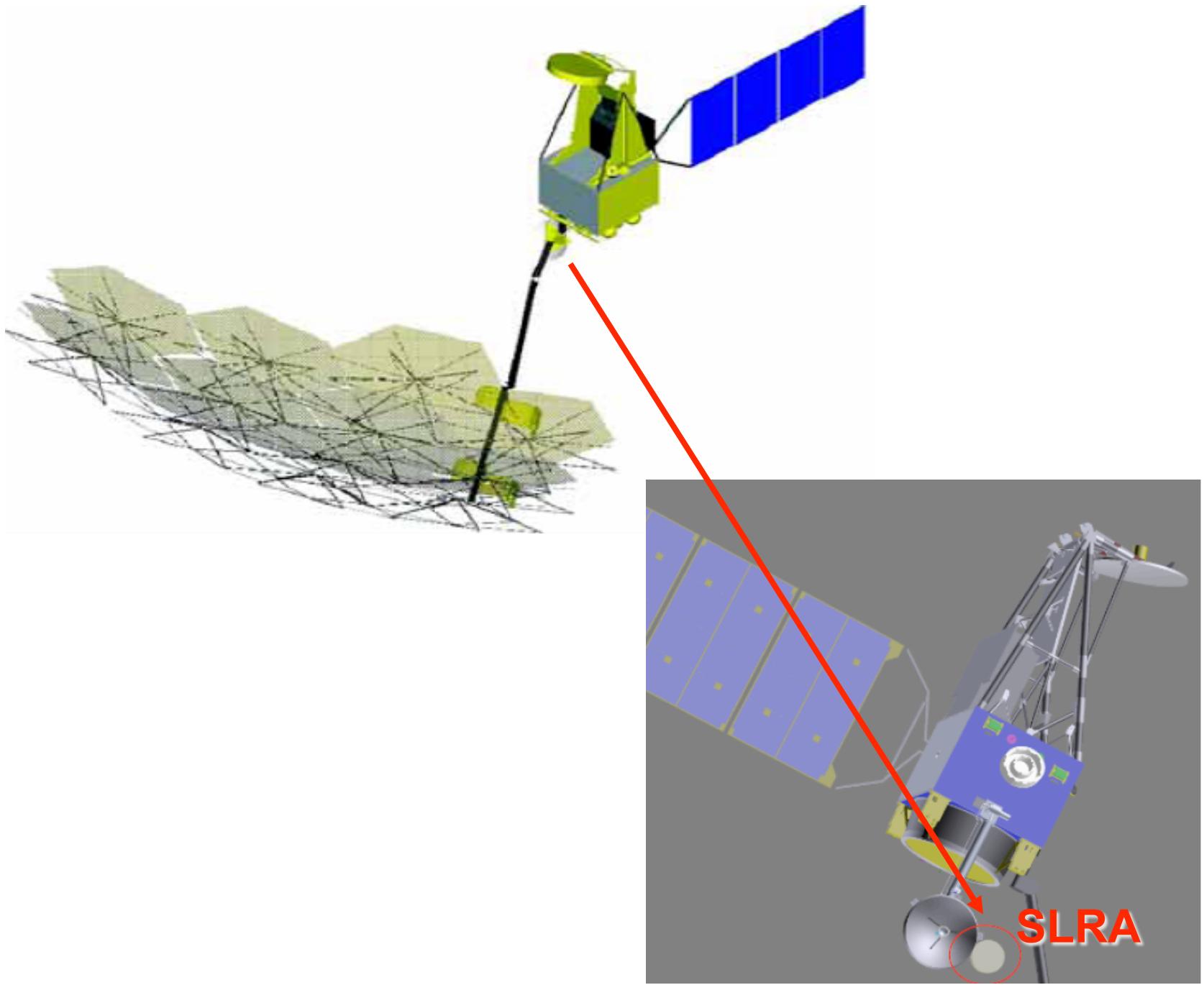
38 mm uncoated CCRs, Dihedral angle = 0.75"

Circular polarisation



**NICT**

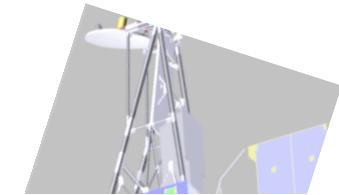
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宇宙航空研究開発機構  
Japan Aerospace Exploration Agency



# Problem 1: Angle of Incidence

## Orientation of ASTRO-G

9.6-m antenna points stars  
by changing the satellite attitude



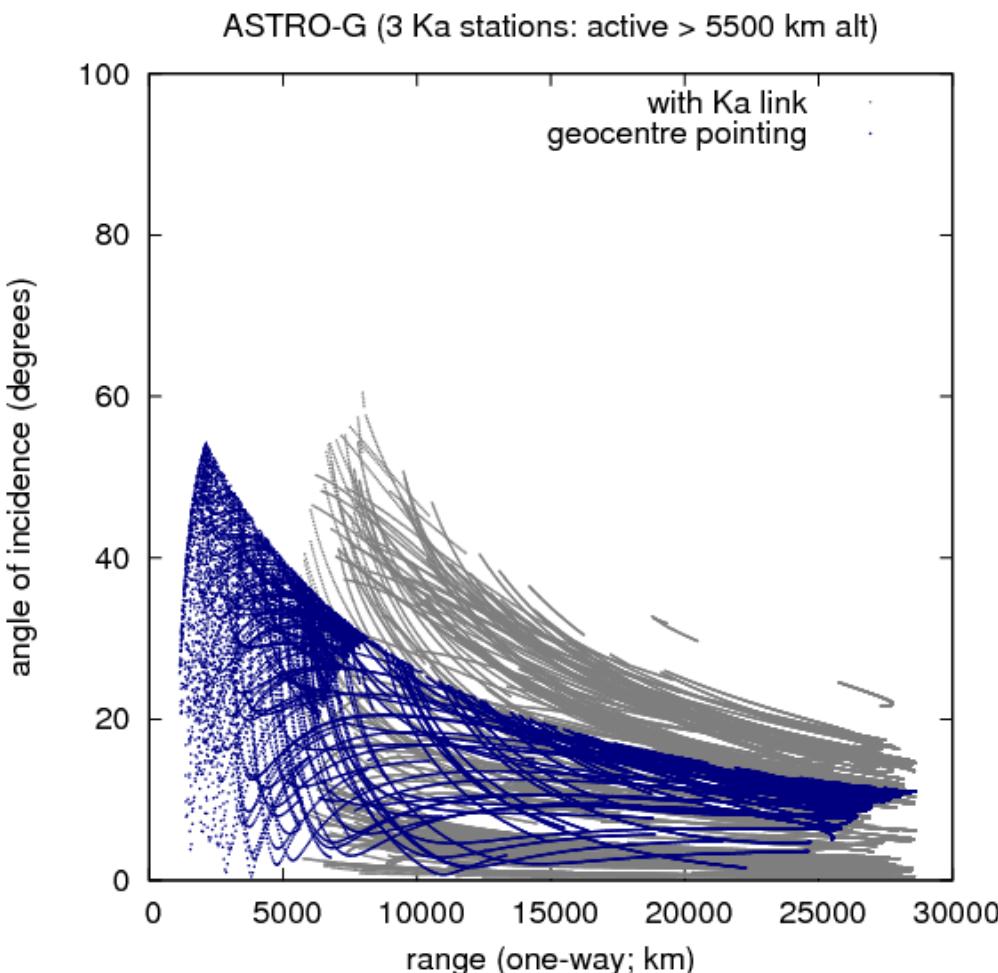
### Ka antenna points one of Ka stations

only when it is visible ( $> 5$  degrees)  
above 5500 km

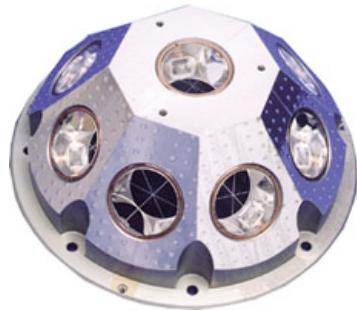
Otherwise,

### Ka antenna points the geocentre

SLR array sync. with Ka antenna



# Reflector Array for LEO and GNSS



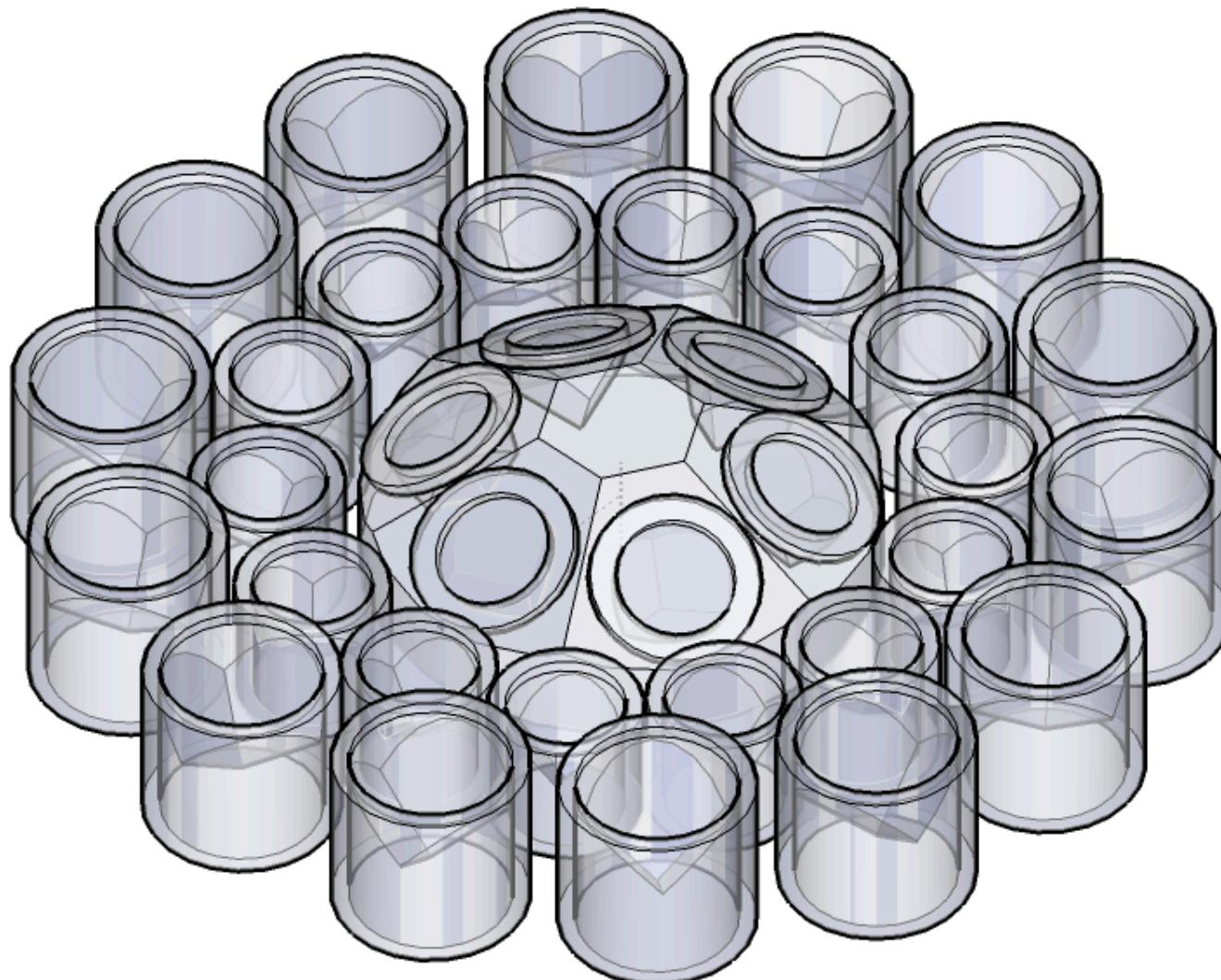
**ALOS**  
*720 km circular orbit*



**GPS**  
*20000 km circular orbit*

Retro pictures: © ILRS Web

# Reflector Array: Basic Design



## Centre:

6 x

28 mm coated CCRs  
Slanted by 30 deg

## Inner Ring:

14 x

28 mm uncoated CCRs

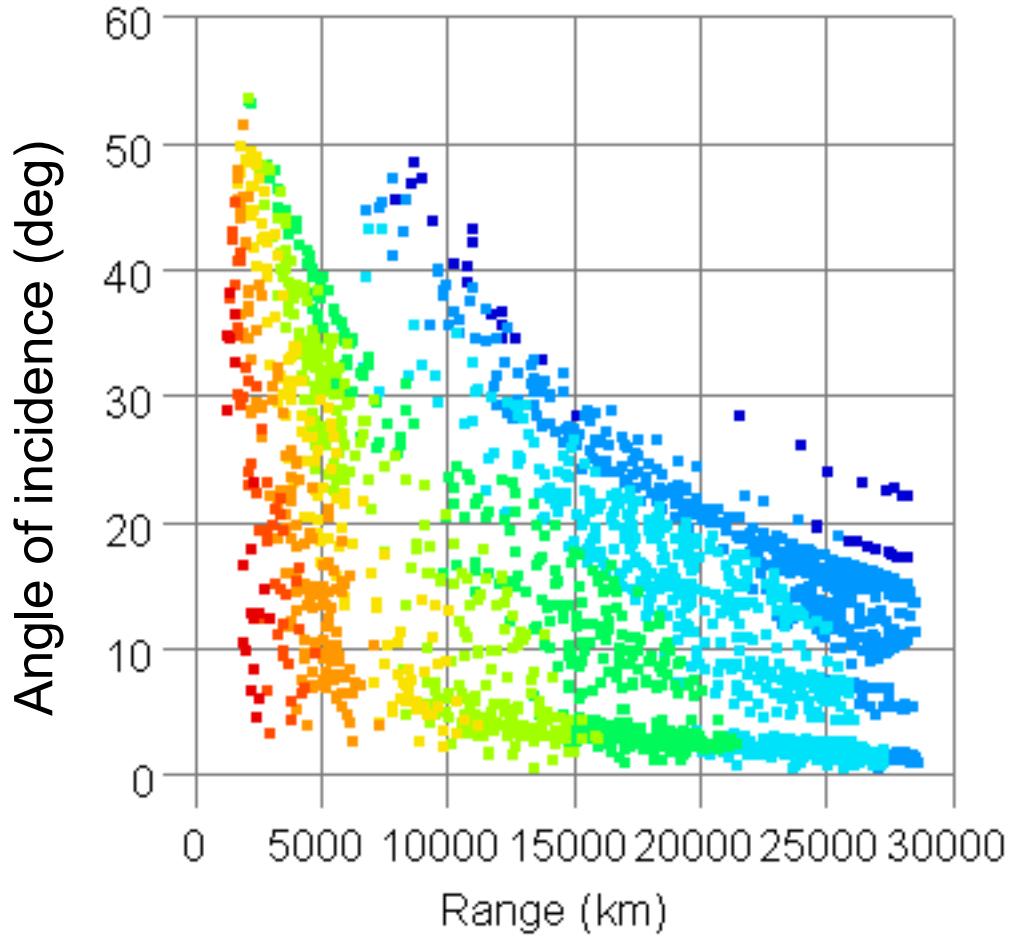
## Outer Ring:

14 x

38 mm uncoated CCRs



# Intensity: ASTRO-G vs ETS-8



Intensity (vs ETS-8)  
 $1.0e+4$

**Time ratio**

**3.2 %**

$\leftarrow LAGEOS \rightleftharpoons$

**18.4 %**

$\dashrightarrow$

**76.2 %**

$\leftarrow ETS-8 \rightleftharpoons$

**2.2 %**

ETS-8 intensity: assuming a 30.5 N station (Tanegashima)

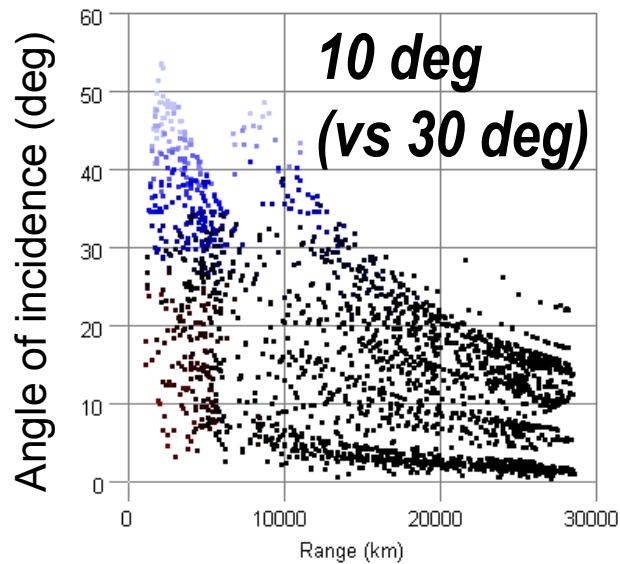


**NICT**

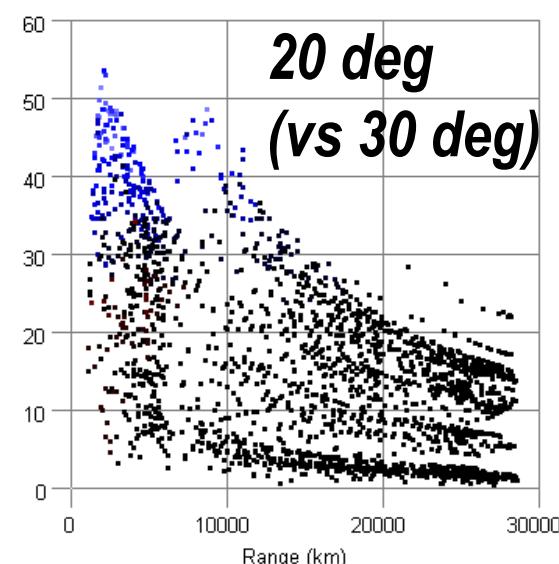
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# Slant angle of centre reflectors

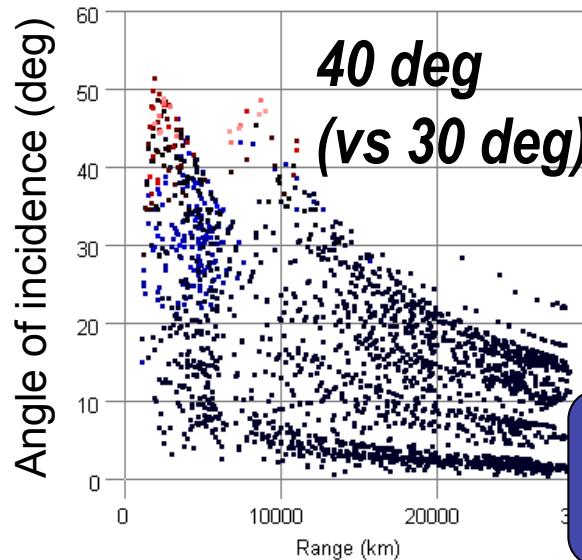
Nominal  
30 deg



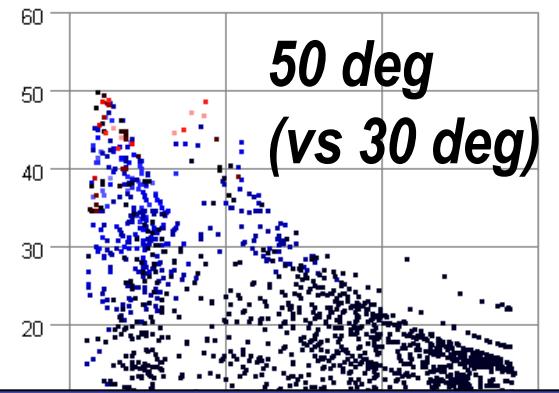
**10 deg**  
(vs 30 deg)



**20 deg**  
(vs 30 deg)



**40 deg**  
(vs 30 deg)



**50 deg**  
(vs 30 deg)

Best slant angle: **30 ~ 40 deg**

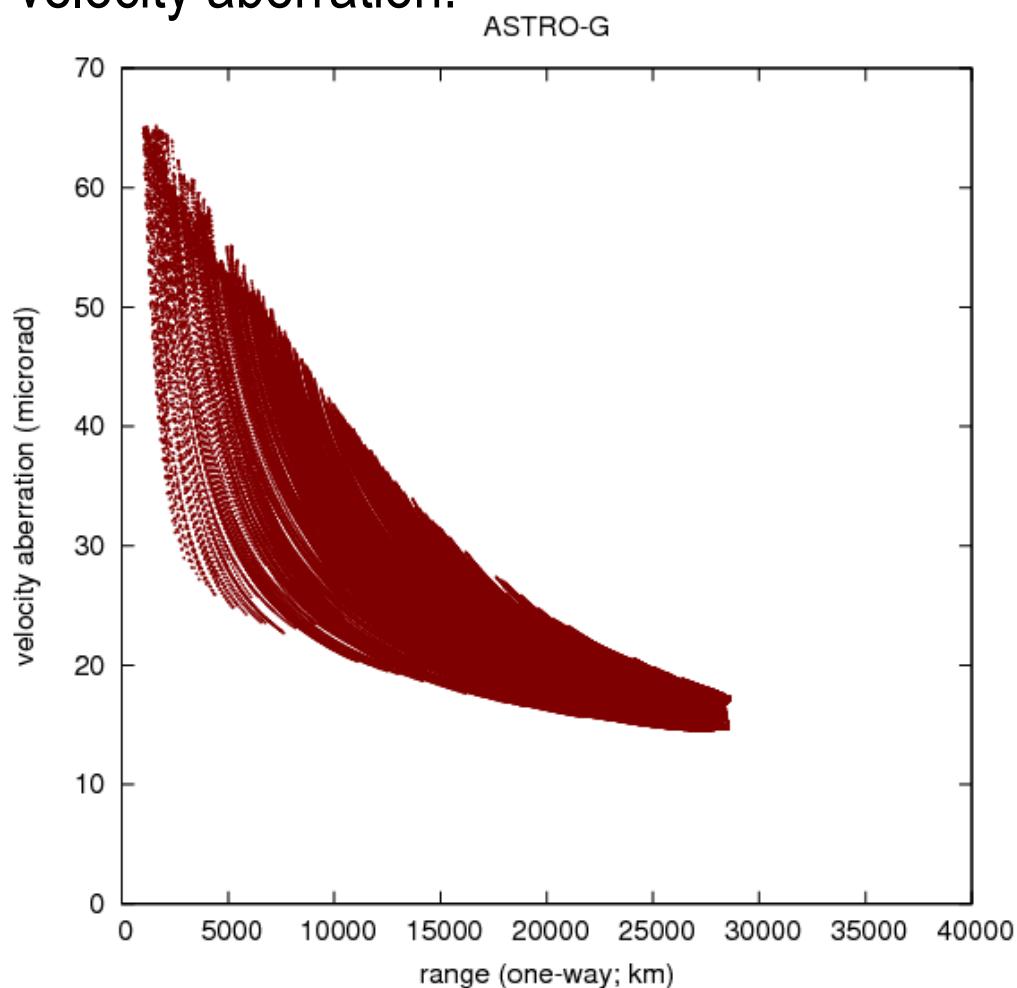
# Problem 2: Velocity Aberration

## Large Velocity Variation of ASTRO-G

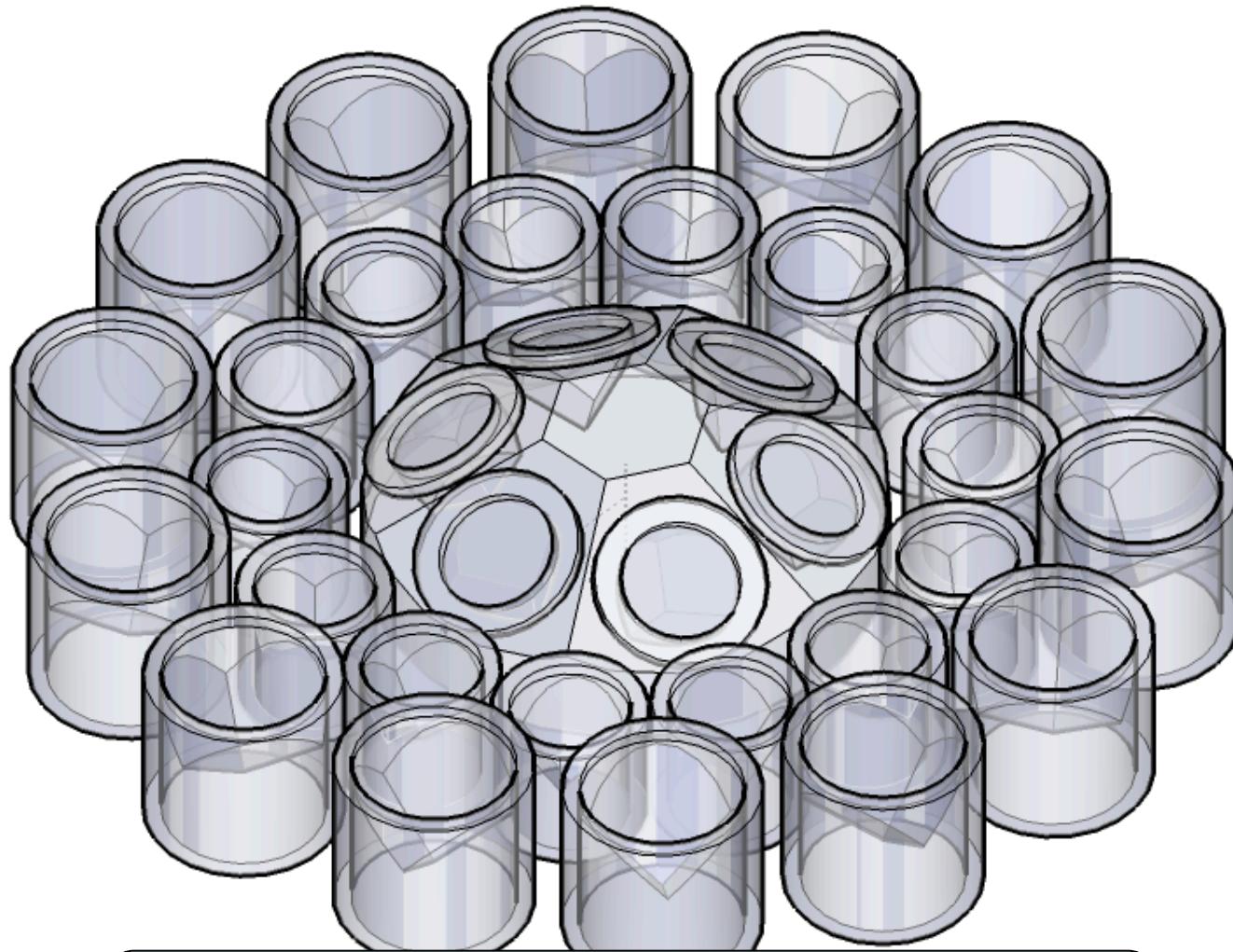
2.2 km/s (apogee) to 9.4 km/s (perigee)

Unprecedented wide range of velocity aberration!

15 to 65 microrad



# Reflector Array: Dihedral Angle



cf. Dihedral angle

~ 1.25" (LAGEOS)  
~ 0.50" (ETS-8)

**Centre:** Dihedral angle  
~ 2.0"  
6 x 28 mm coated CCRs  
Inclined by 30 deg

**Inner Ri** Dihedral angle  
< 0.75"  
14 x 28 mm uncoated CCRs

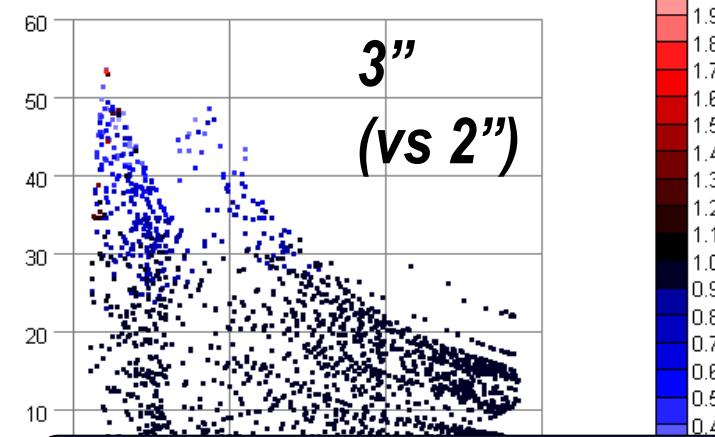
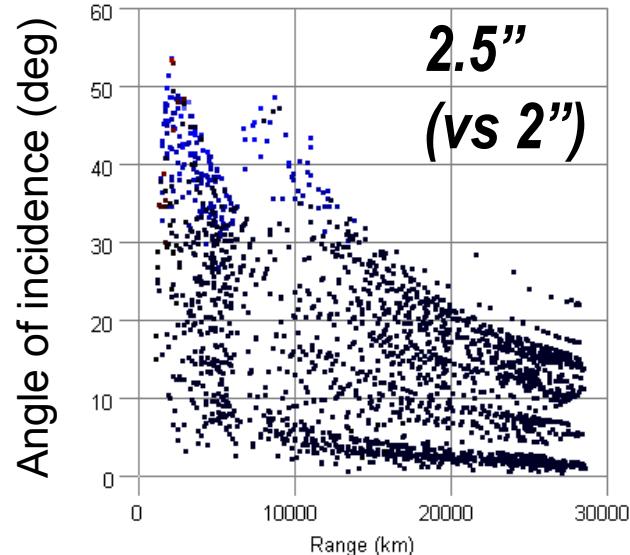
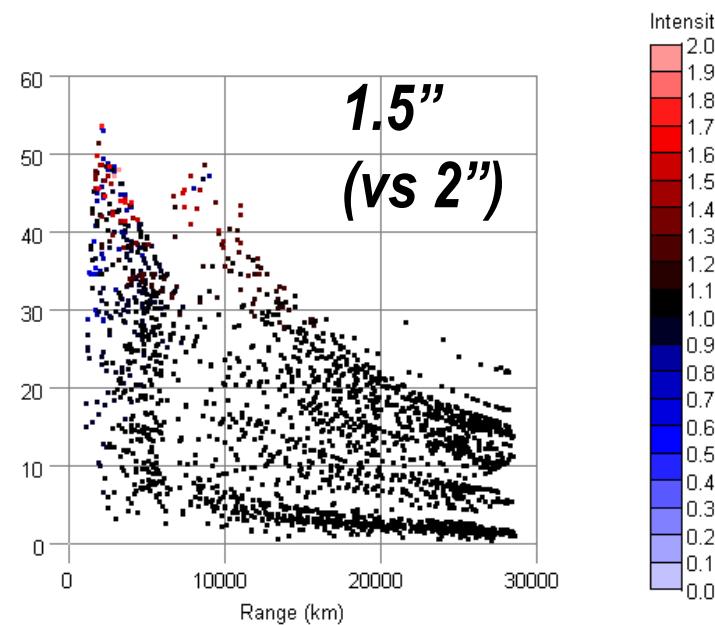
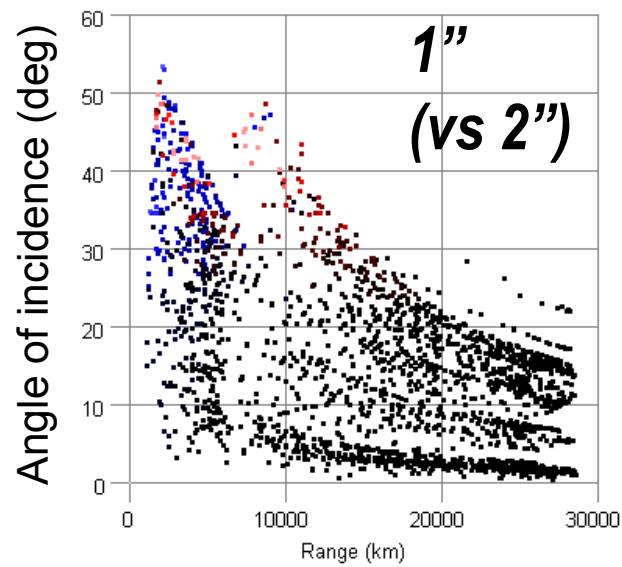
**Outer Ri** Dihedral angle  
< 0.75"  
14 x 38 mm uncoated CCRs



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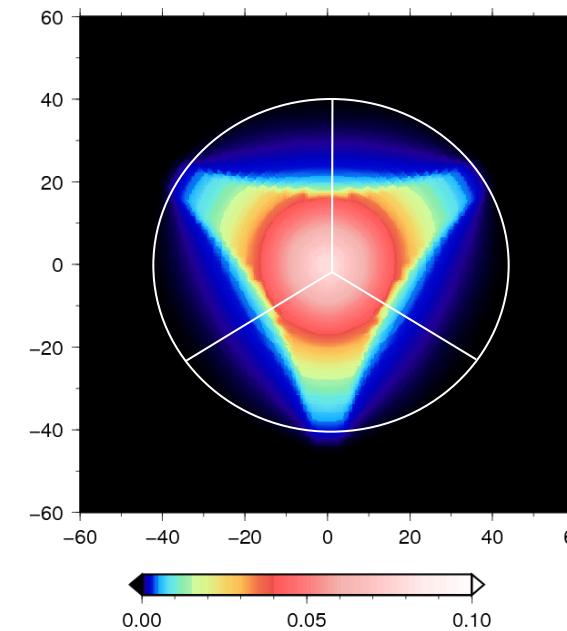
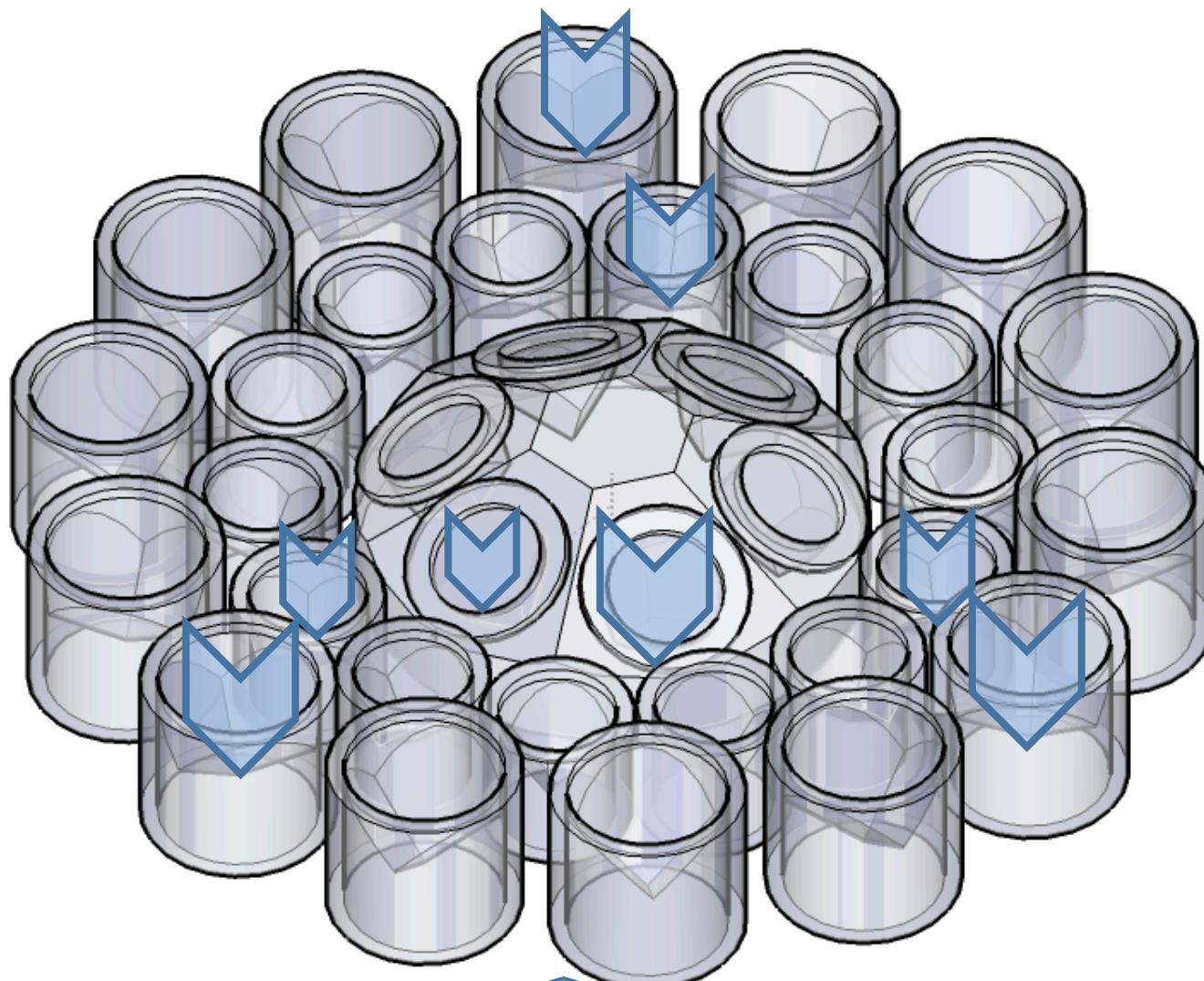
# Dihedral angle of centre reflectors

Nominal  
2.0"



Best dihedral angle: 1.5" ~ 2"

# Reflector Array: Orientation & Double Pulse

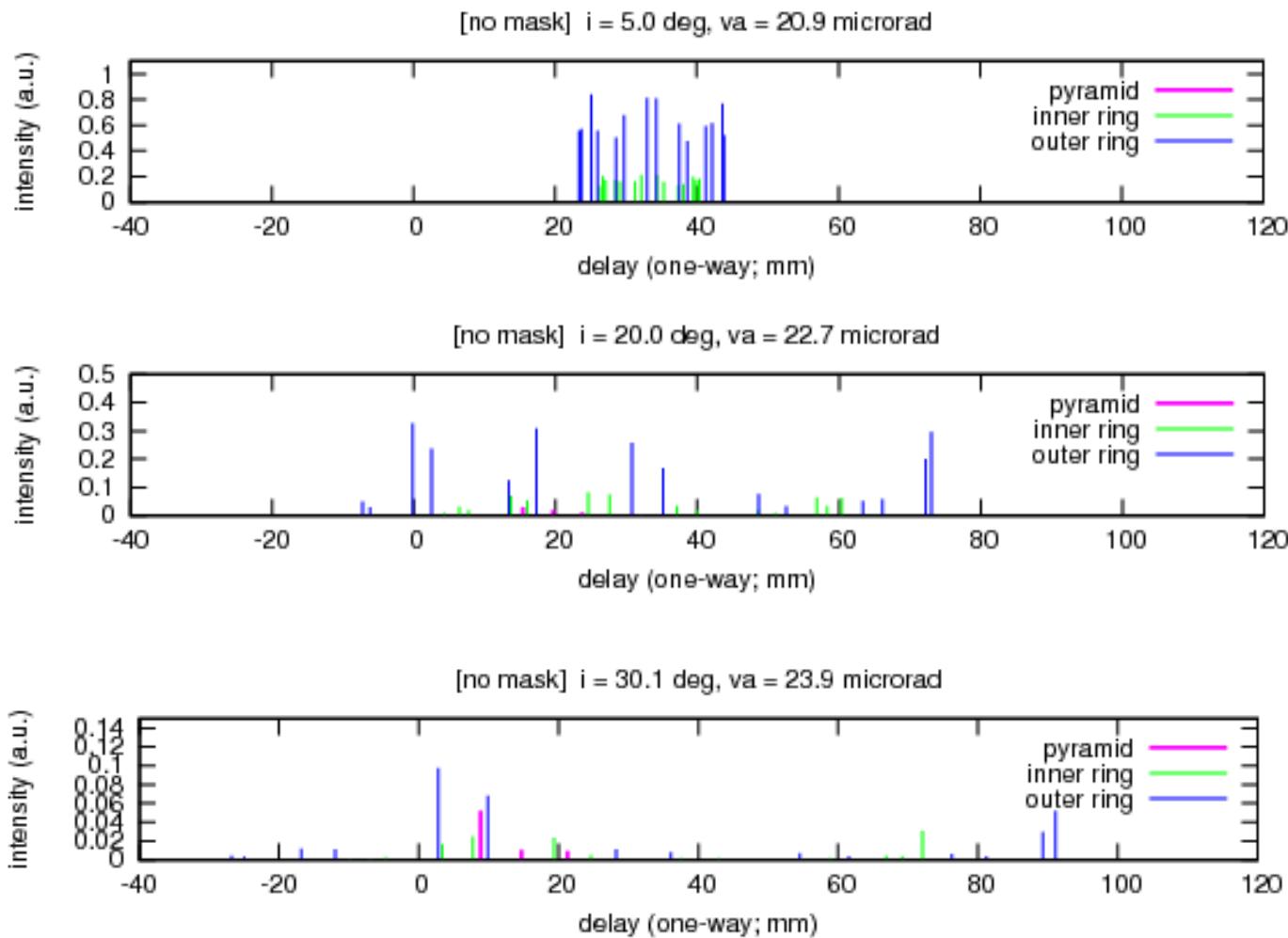


Single Reflector  
(uncoated)

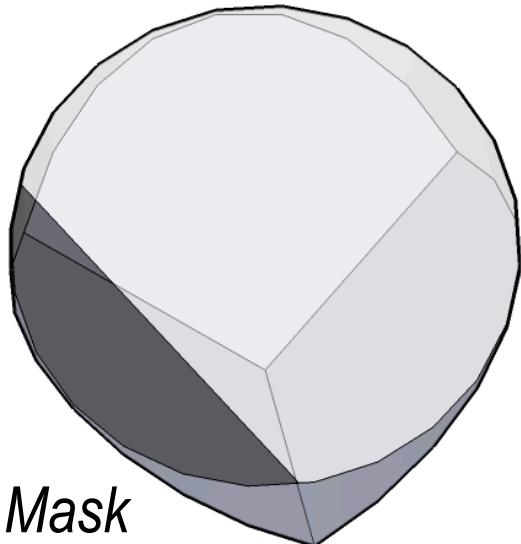
Strong 120-deg  
az-dependence



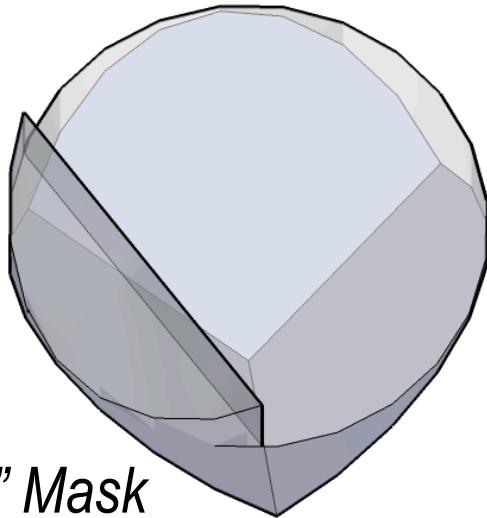
# Problem 3: Double pulse



# Mask Ideas

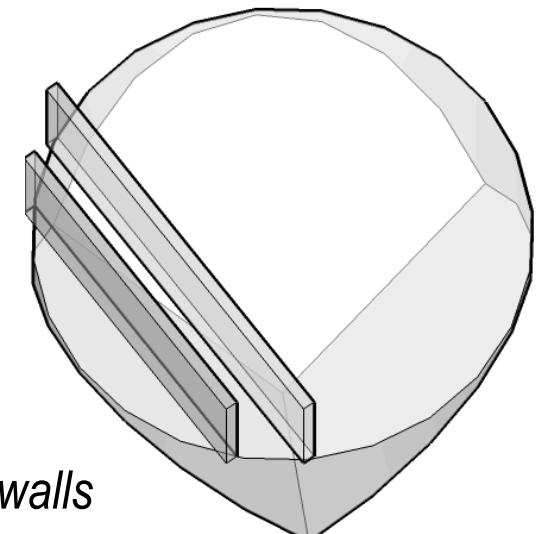


“Flat” Mask



“Delta” Mask

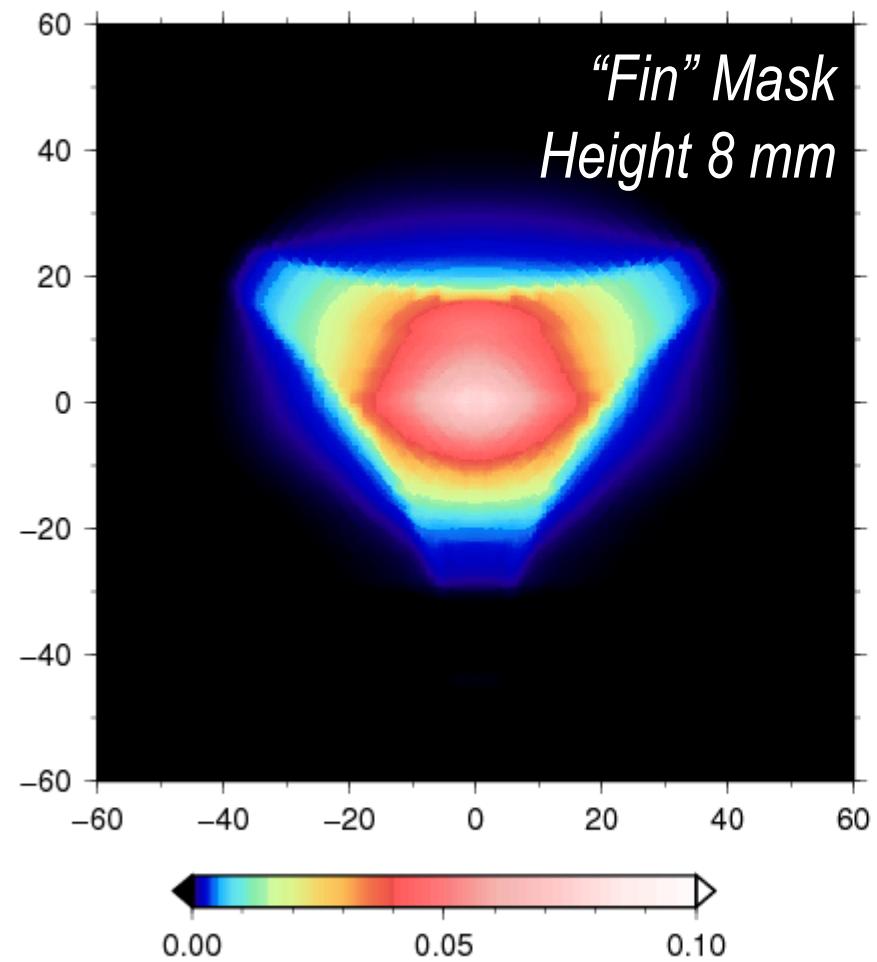
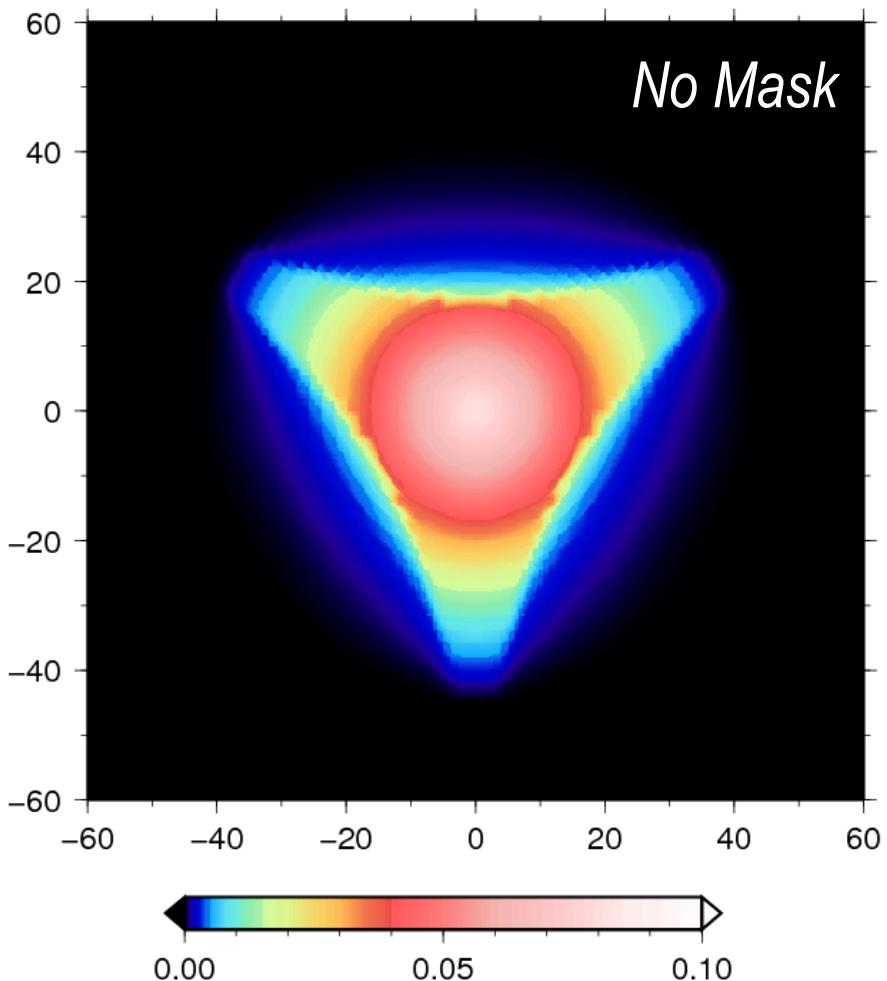
Precision  
vs  
Intensity



“Fin” Mask

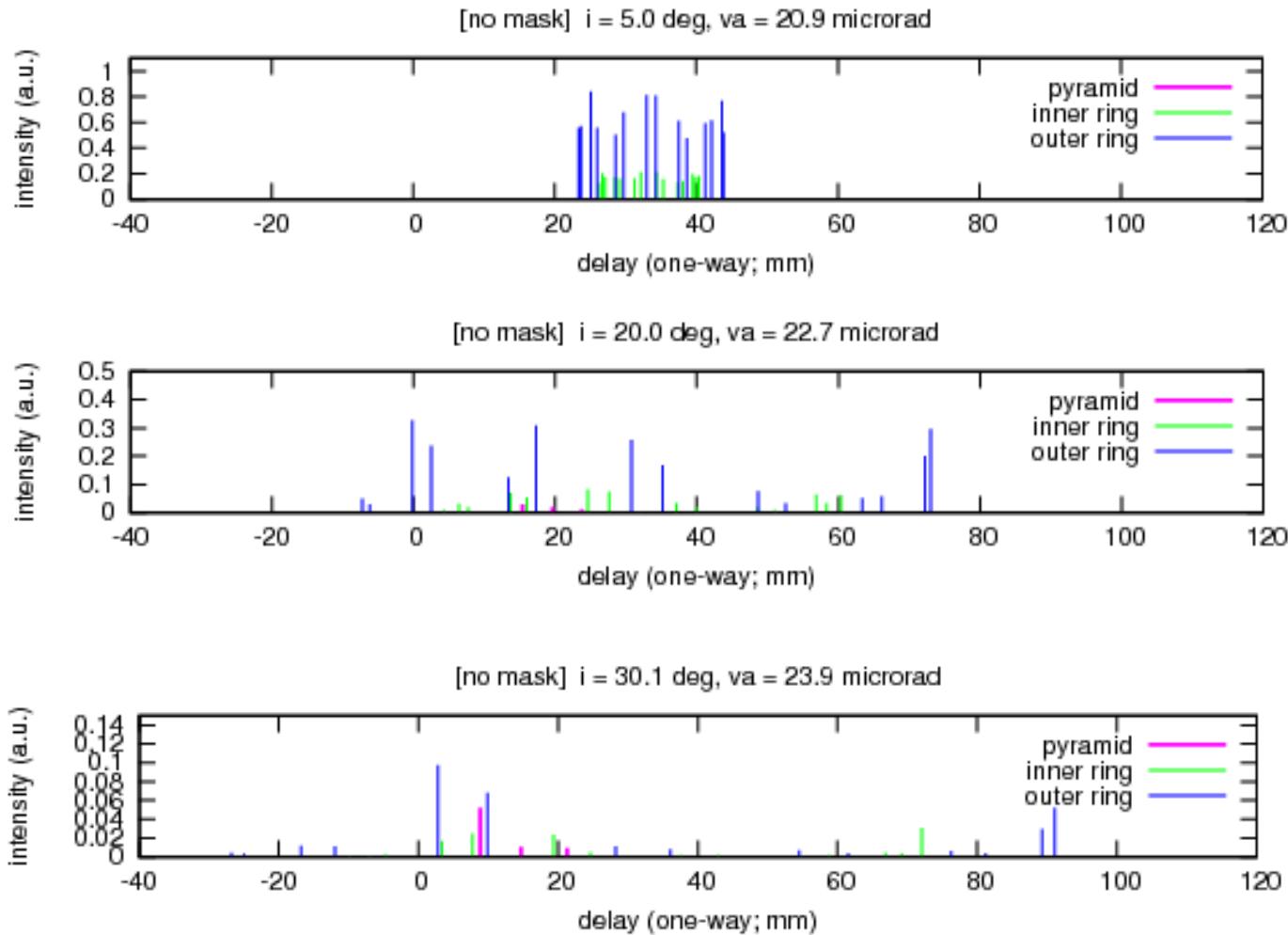
Two  $x$ -mm height walls  
at  $0.25 r$  and  $0.5 r$

# Effect of masks



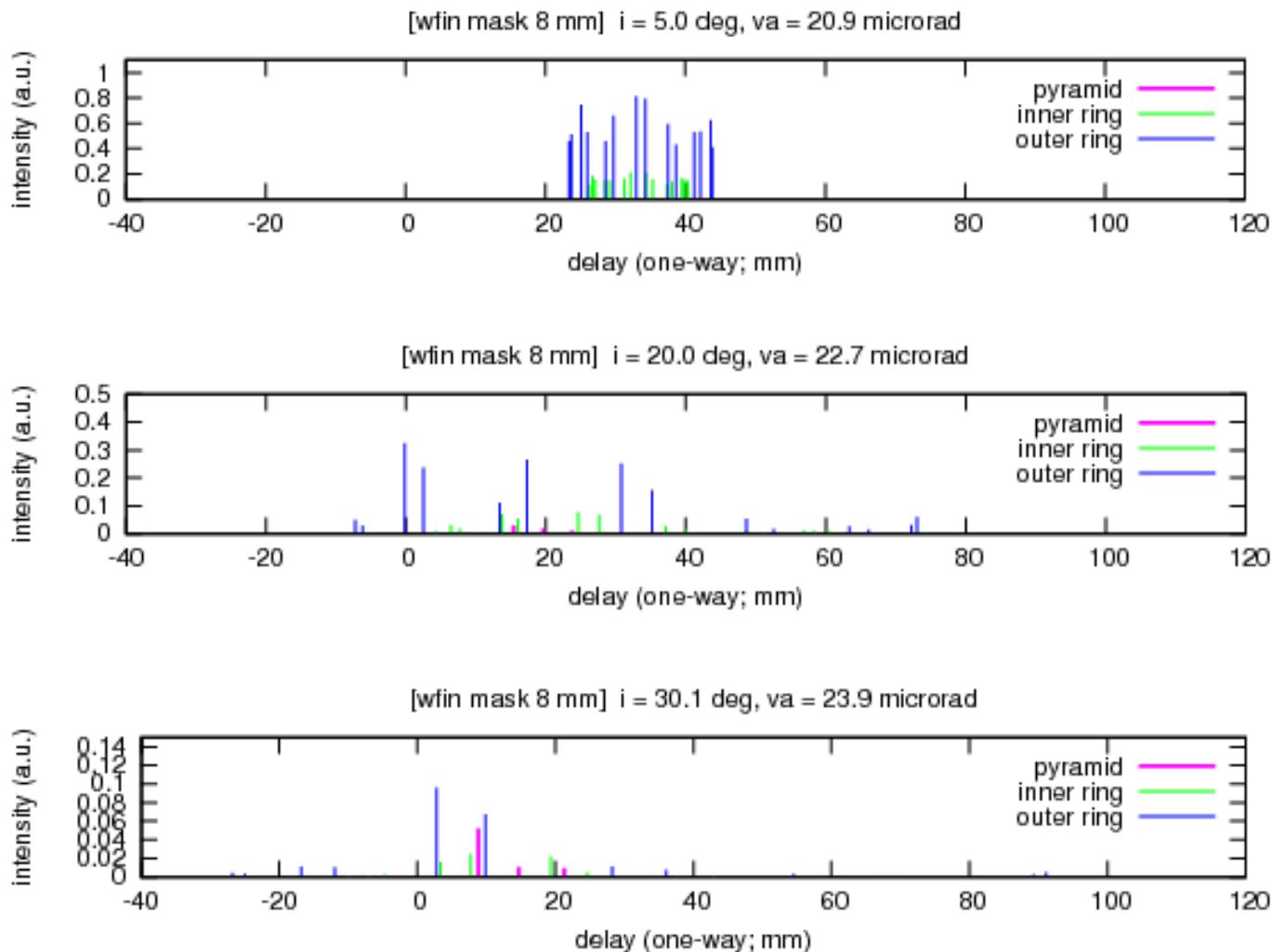
No  
Mask

# Problem 3: Double pulse



# Double pulse eliminated!

Fin  
8 mm  
Height



# Summary

## Array design

Mixture of LEO-type and GNSS-type

Return Intensity: Mostly stronger than ETS-8, Huge variation in time

## Plenty of new problems/concepts due to its orbit

Angle of incidence

Velocity aberration

Double pulse elimination

## Basic design done, but still needs fine tuning

**We sincerely hope your supports to this challenging target.**

*Laser ranging helps blackhole studies for the first time ever!*

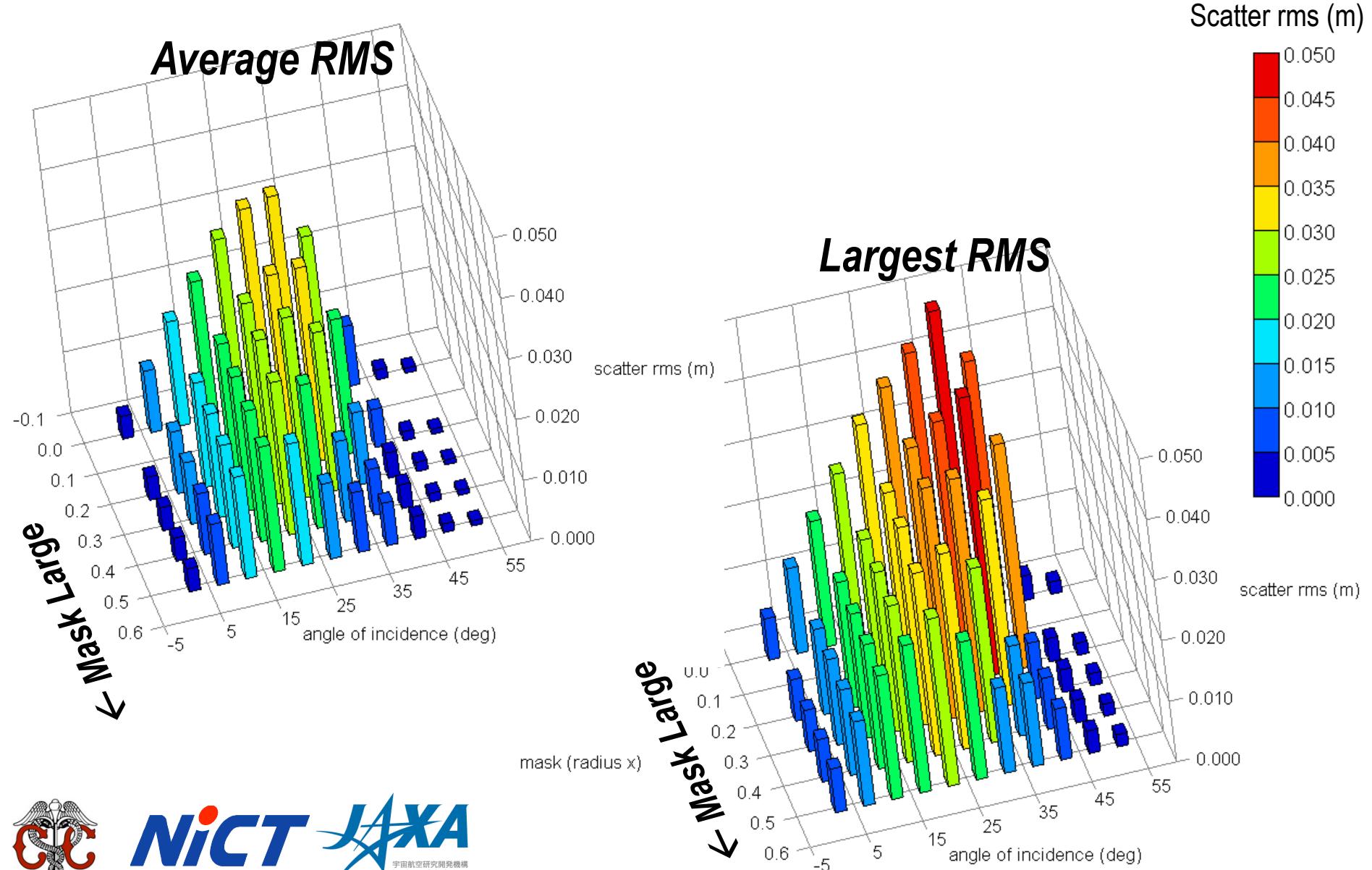
*(But its launch is FY2012 = still 4 years ahead)*





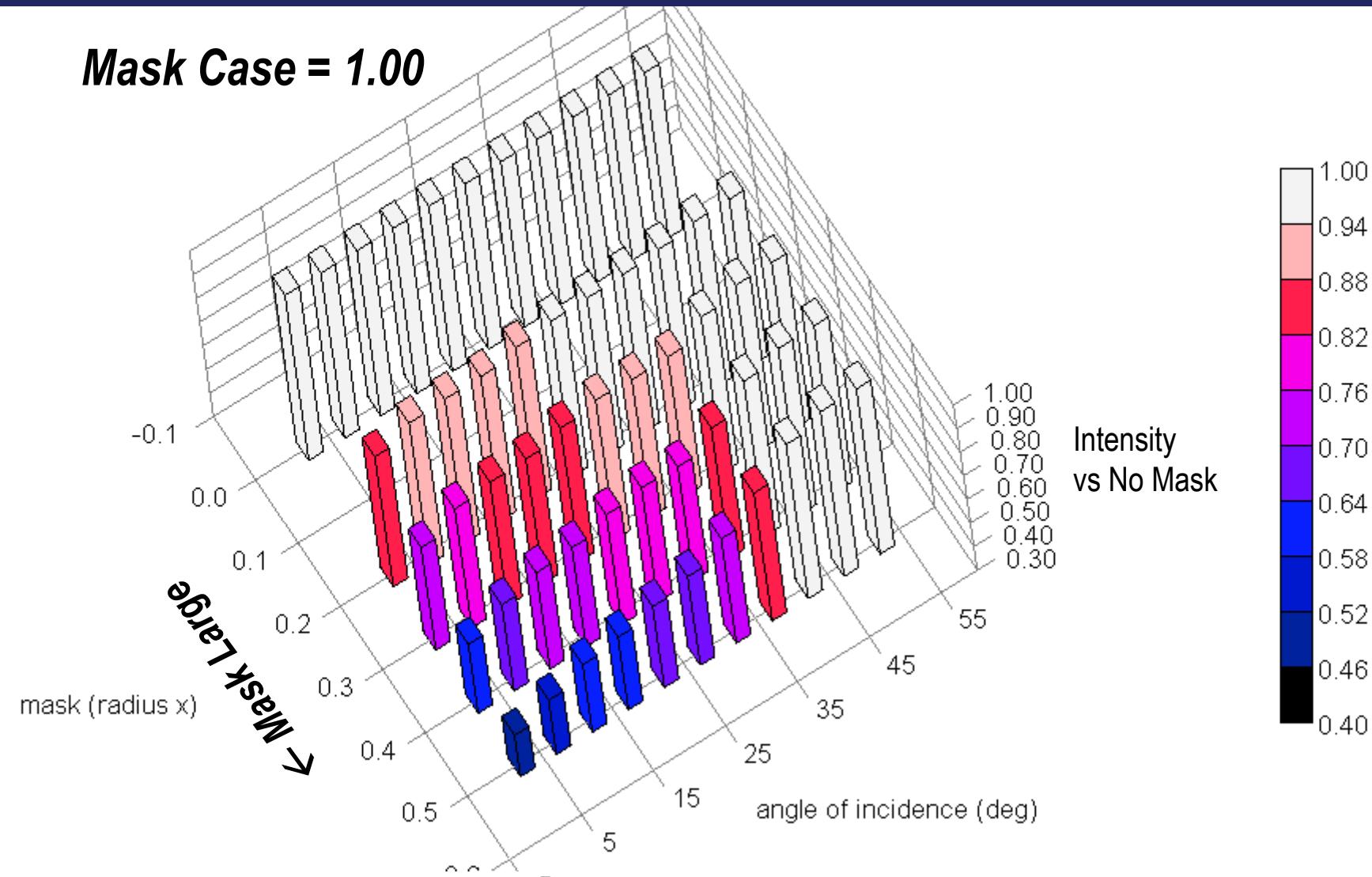


# Flat mask cases: Scatter RMS

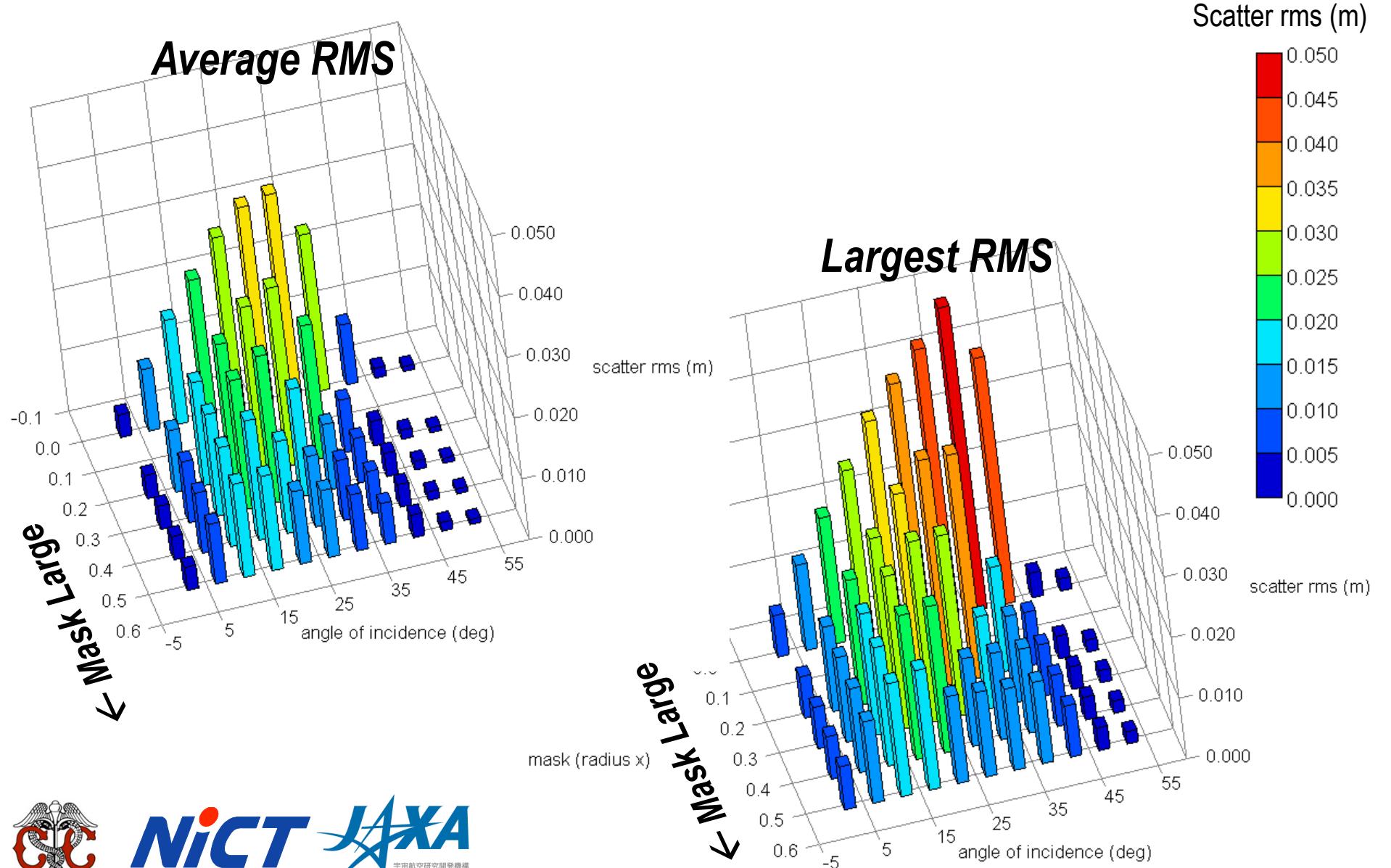


# Flat mask cases: Intensity

**Mask Case = 1.00**

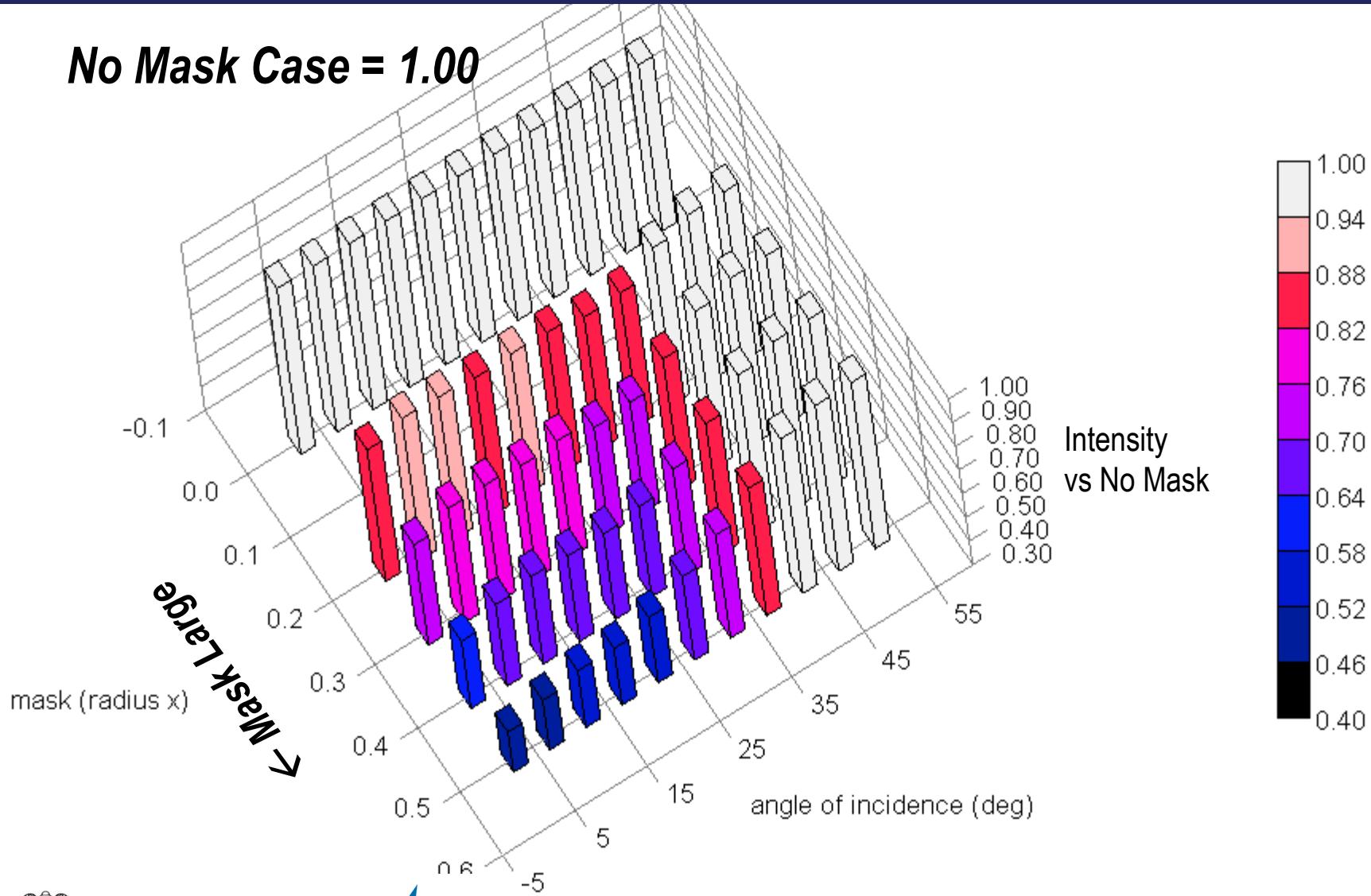


# Delta mask cases: Scatter RMS

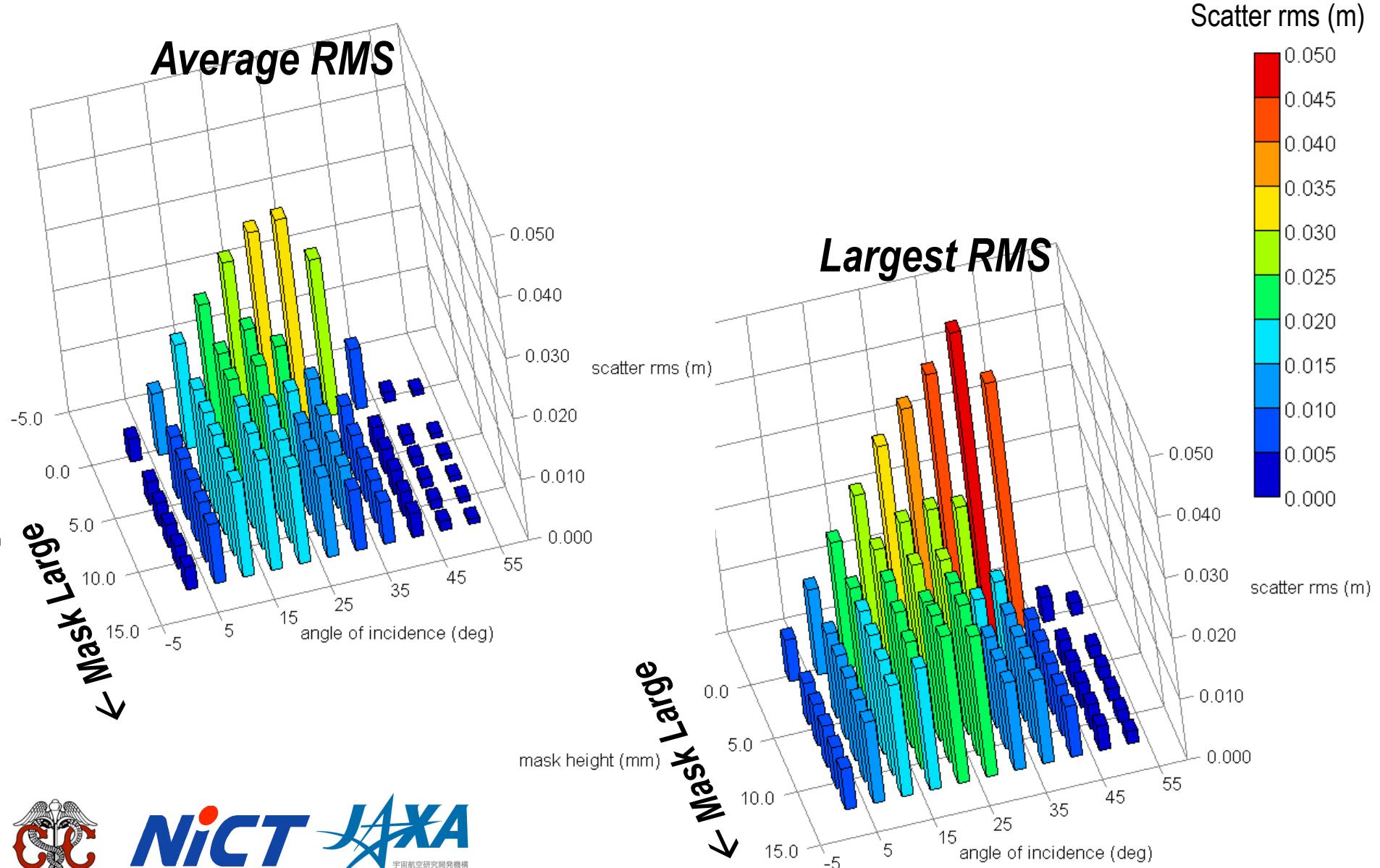


# Delta mask cases: Intensity

No Mask Case = 1.00

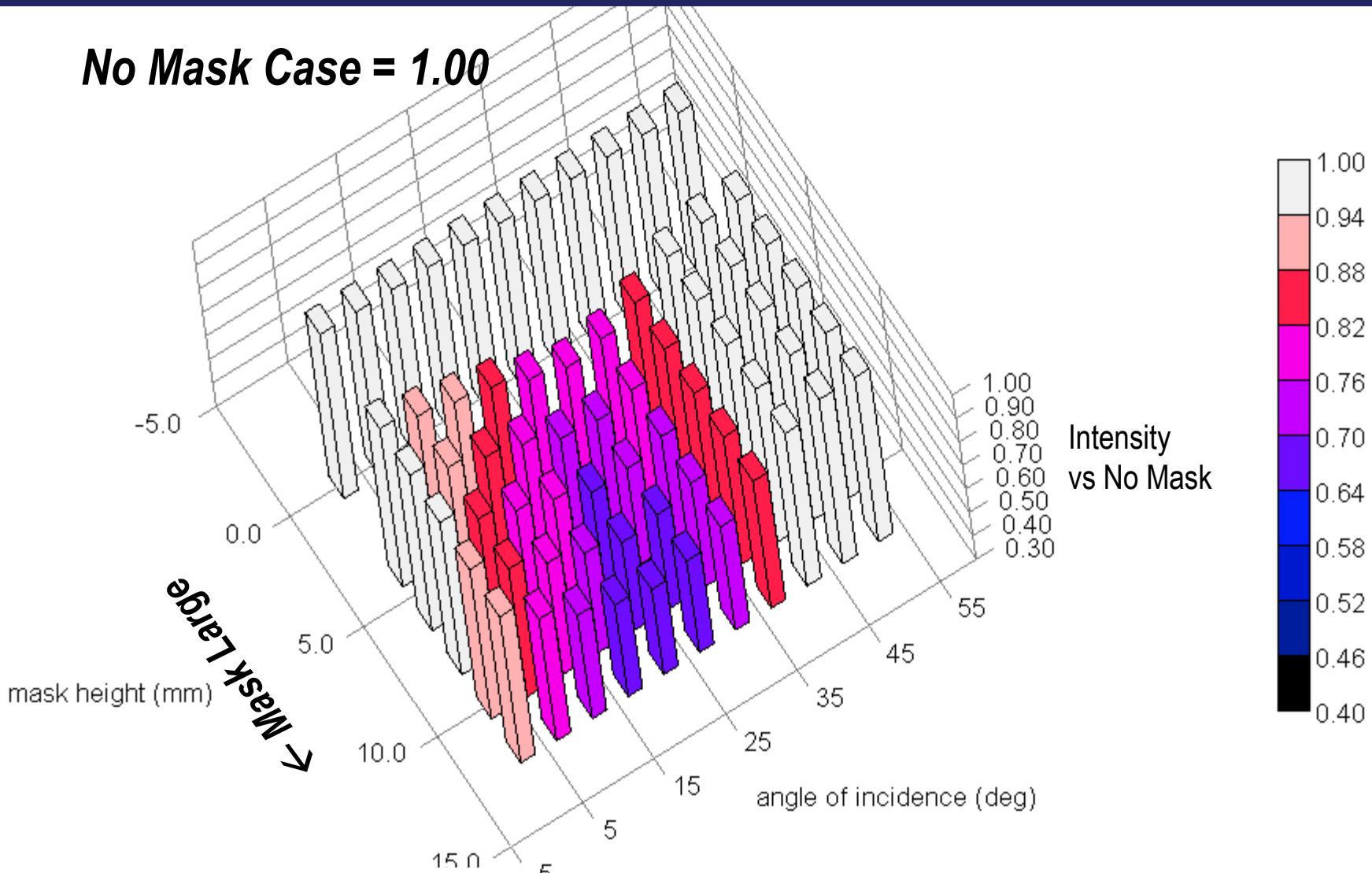


# Fin mask cases: Scatter RMS



# Fin mask cases: Intensity

No Mask Case = 1.00



Intensity  
vs No Mask